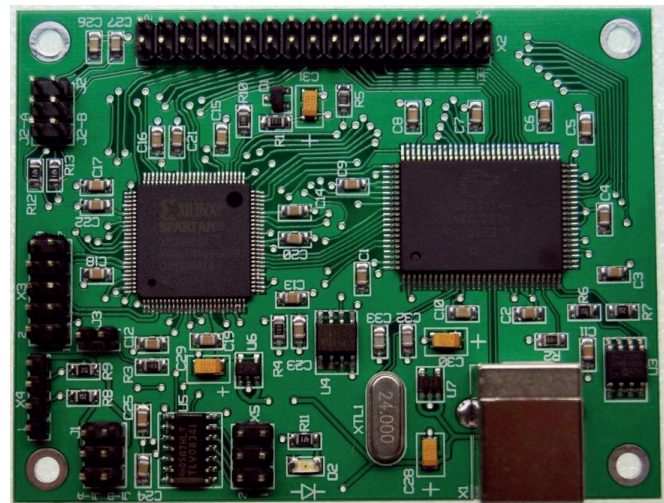


# Control system for the CubeSat micropropulsion module @ SPACE-SI

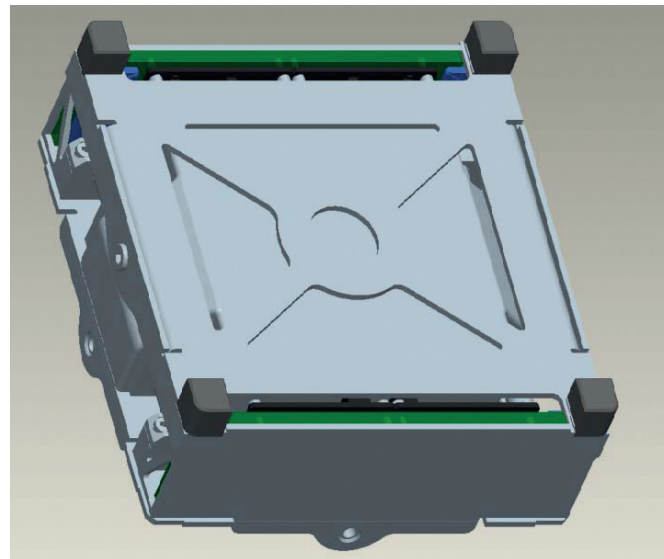
In the Slovenian Centre of Excellence Space-SI we are developing a control system for the CubeSat micropropulsion module designed by SSC NanoSpace.

Micropropulsion module includes four MEMS (Micro Electro Mechanical Systems) chips and ON/OFF valves. Each MEMS chip contains flow control valve, flow sensor and chamber/nozzle together with the front-end read-out electronics.

The control system is implemented as a separate printed circuit board and serves as a closed-loop control system for the micro-thrusters. It also provides a suitable communication interface to the other modules of the satellite, e.g., on-board computer (OBC) and attitude determination control system.



Controller board	
Volume	10x10x2 cm <sup>3</sup>
Mass	80 g
Power Consumption	500 mW (average, operating)
Supply voltage	5 V, 12 V
Communication interface	Via I2C, CAN or RS485 bus
CubeSat propulsion module	
Thrusters	four 1mN thrusters with 10 <sup>0</sup> N resolution
Total impulse	40 Ns
Weight	250 g (dry)
Propellant capacity	50 g (butane)
Operating pressure	2-5 bar
Power consumption	2W (average, operating)
Electrical interface	52 pins analog (0-12V) and digital (SPI)



The circuit of the interface board consists of a microcontroller with corresponding peripherals, such as A/D, D/A-PWM converters and power amplifiers for actuator signals. The outputs from the interface board are electrical signals to actuate the MEMS valves and the ON/OFF valves of the propulsion system.

The controller board controls mass flow through each of the four MEMS thrusters with respect to the desired thrust, which is demanded by the OBC over a data bus. Also the closed-loop's parameters can be changed during the mission and the parameters, which describe the state of the propulsion system, can be transmitted to the OBC.

